

United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number SE74WE

This certificate issued to Lewis Lalbee

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 13 of the Civil Air Regulations, effective June 15, 1956, as amended by 13-1 through 13-4.

Original Product — Type Certificate Number: E7CE
Make: Continental
Model: GTS10-520-G, -D

Description of Type Design Change:

BG-341 spark plug as eligible for installation on the above Continental engines.

Limitations and Conditions:

Spark plugs must be visually inspected for obvious defects prior to their installation in accordance with normal aircraft maintenance practice. Included in this inspection must be a check of the electrode gap. The proper gap for this plug is .014".

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: December 27, 1974

Date issued:

Date of issuance: January 28, 1975

Date amended:



By direction of the Administrator
(Signature)

Acting Chief, Aircraft Engineering Division
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

United States of America
Department of Transportation - Federal Aviation Administration
Supplemental Type Certificate

Number SE69WE

This certificate issued to Lewis Laibee

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified herein meets the airworthiness requirements of Part 13 of the Civil Air Regulations, effective June 15, 1956, as amended by 13-1.

Original Product - Type Certificate Number: 3E1
Make: Continental
Model: IO-470-D, -E, -F, -H, -L, -M, -N, -S, -U, -V, -VO

Description of Type Design Change:
BG-341 spark plug as eligible for installation on the above Continental engines.

Limitations and Conditions:

Spark plugs must be visually inspected for obvious defects prior to their installation in accordance with normal aircraft maintenance practice. Included in this inspection must be a check of the electrode gap. The proper gap for this plug is .014".

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: December 27, 1974

Date issued:

Date of issuance: January 28, 1975

Date amended:



By direction of the Administrator
[Signature]
(Signature)

Acting Chief, Aircraft Engineering Division
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.



United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number SE73WE

This certificate, issued to Lewis Leibee

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 13 of the Civil Air Regulations, effective June 15, 1956, as amended by 13-1 through 13-6.

Original Product — Type Certificate Number: E8CE

Make: Continental

Model: TS10-520-A, -B, -C, -D, -E, -G, -H, -J, -K, -L

Description of Type Design Change:

EG-341 spark plug as eligible for installation on the above Continental engines.

Limitations and Conditions:

Spark plugs must be visually inspected for obvious defects prior to their installation in accordance with normal aircraft maintenance practice. Included in this inspection must be a check of the electrode gap. The proper gap for this plug is .014".

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: December 27, 1974

Date received:

Date of issuance: January 28, 1975

Date amended:



By direction of the Administrator
[Signature]
(Signature)

Acting Chief, Aircraft Engineering Division
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.



United States of America
Department of Transportation—Federal Aviation Administration
Supplemental Type Certificate

Number SE72WE

This certificate, issued to Lewis Leibee

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part 13 of the Civil Air Regulations, effective June 15, 1956, as amended by 13-1 through 13-5.

Original Product—Type Certificate Number: E5CE

Make: Continental

Model: IO-520-A, -B, -BA, -C, -D, -E, -F, -J, -K,
-L, -M

Description of Type Design Change:

BC-341 spark plug as eligible for installation on the above Continental engines.

Limitations and Conditions:

Spark plugs must be visually inspected for obvious defects prior to their installation in accordance with normal aircraft maintenance practice. Included in this inspection must be a check of the electrode gap. The proper gap for this plug is .014".

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: December 27, 1974

Date issued:

Date of issuance: January 28, 1975

Date amended:



By direction of the Administrator
OK Sample
(Signature)

Acting Chief, Aircraft Engineering Division
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.